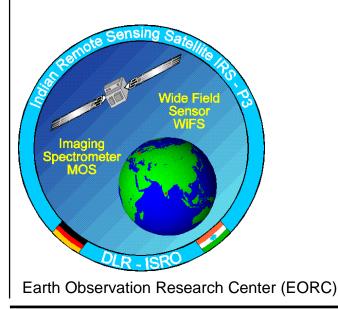
On-board Calibration MOS-IRS and GLI

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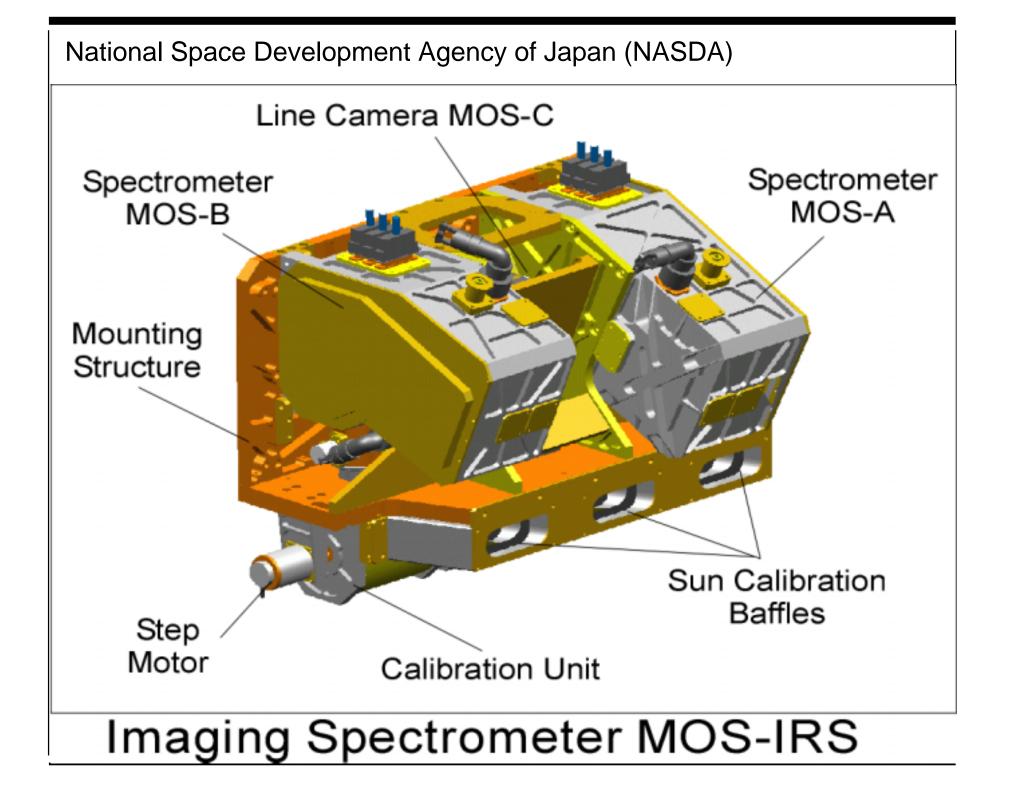


On-board calibration comparison

- 1. MOS-IRS, an experimental coastal zone imaging spectrometer
- 2. MOS-IRS after 4.5 years of on-board calibration
- 3. GLI, a full operative global orientated whiskbroom scanner
- 4. GLI on-board calibration
- 5. Comparison GLI MOS-IRS calibration



6. Conclusion



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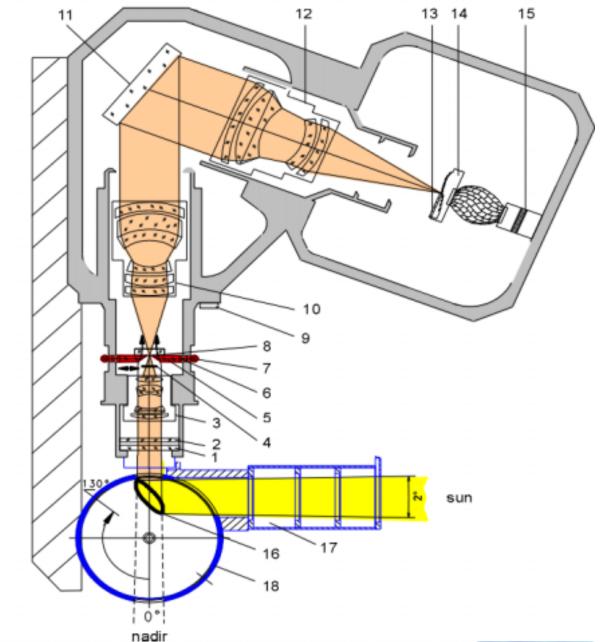
Modular Optoelectronic Scanner MOS - Main Parameters

Parameter	MOS-A	MOS-B	MOS-C
Spectral Range [nm]	755 - 768	408 - 1010	SWIR
No. of Channels	4	13	1
Wavelengths [nm]	756.7; 760.6; 763.5; 766.4 O ₂ .A-band	408; 443; 485; 520; 570; 615; 650; 685; 750; 870; 1010 815; 945 (F ₂ O-vapor)	1600
spectral halfwidth [nm]	1.4	10	100
FOV along track x [deg] across track [deg]	0.344 13.6	0.094 14.0	0.14 13.4
Swath Width [km]	195	200	192
No. of Pixels	140	384	299
Pixel Size x*y [km ²]	1.57x1.4	0.52x0.52	0.52x0.64
Measuring Range L _{min} I max [µWcm ⁻² nn ⁻¹ sr ⁻¹]	0.140	0.2 65	0.5 18
ΔL/L [%]	0.3	1.0	2.0



MOS - B Optical design with Sun cal unit

- 1 quartz plate
- 2 quarter wave plate
- 3 objective "Tevidon" 1,4/25
- 4 shutter
- 5 glass prism
- 6 filter glass FB 120
- 7 mini lamp
- 8 entrance slit 0,041 mm x 6,140 mm
- 9 adjustment mirror
- 10 collimator "Pancolar" 1,8/80
- 11 grating 325 Vmm
- 12 imager "Pancolar" 2,8/120
- 13 filter glass RG 5
- 14 focal plane assembly
- 15 peltier cooler
- 16 sun diffuser
- 17 buffle
- 18 sun cal unit



Results of MOS-IRS on-board calibration

Lamps: total error: 1.1%; frequency: before each orbit

- relative radiometric, linearity, spectral checks
- show that sensor sensitivity is stable or increases slowly (MOS-B up to +7%)
- no changes of linearity and spectral checks

Sun: total error: 2.0 %; frequency: ca. 15 times a year

- absolute radiometric calibration
- error due to stray light effects in the SunCal Unit (straylight has a strong correlation with seasonal variations of Sun incidence angle)
- no degradation of the diffuser plate

global caudger

(state: April 1996 — October 2000)

After 4.5 years of MOS-IRS on-board calibration

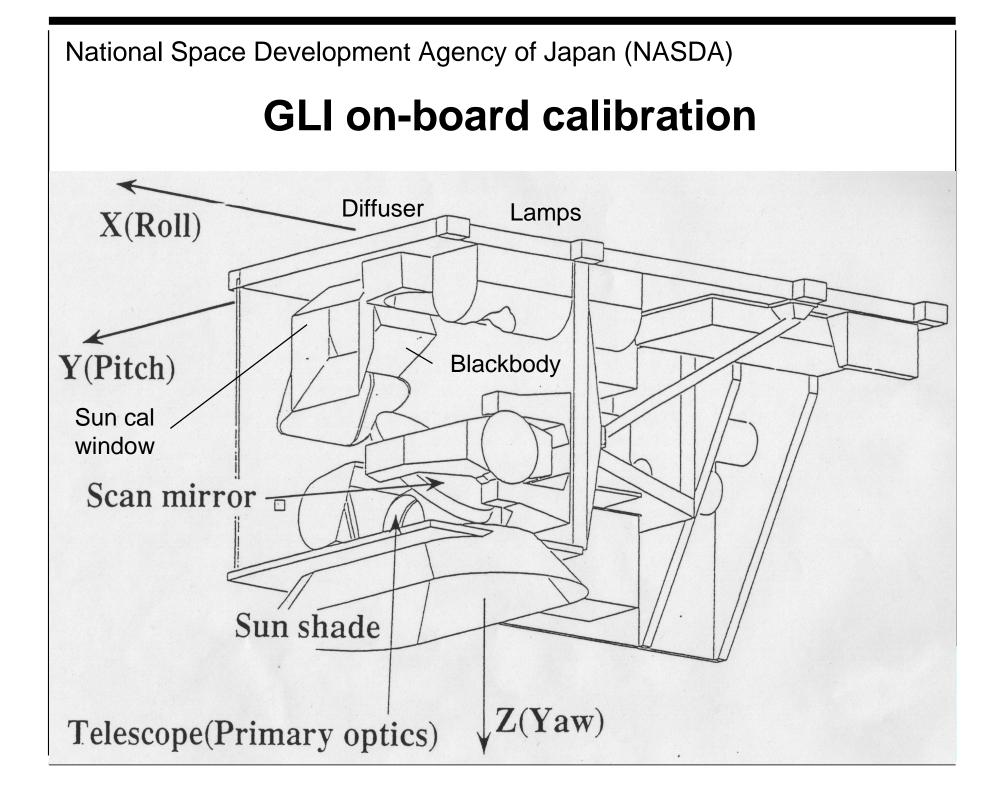
- Sun calibration is a reliable source for absolute calibration
- Spectralon is a very suitable and stable material for in-orbit Sun calibration
- lamp calibration is a reliable source for relative calibration

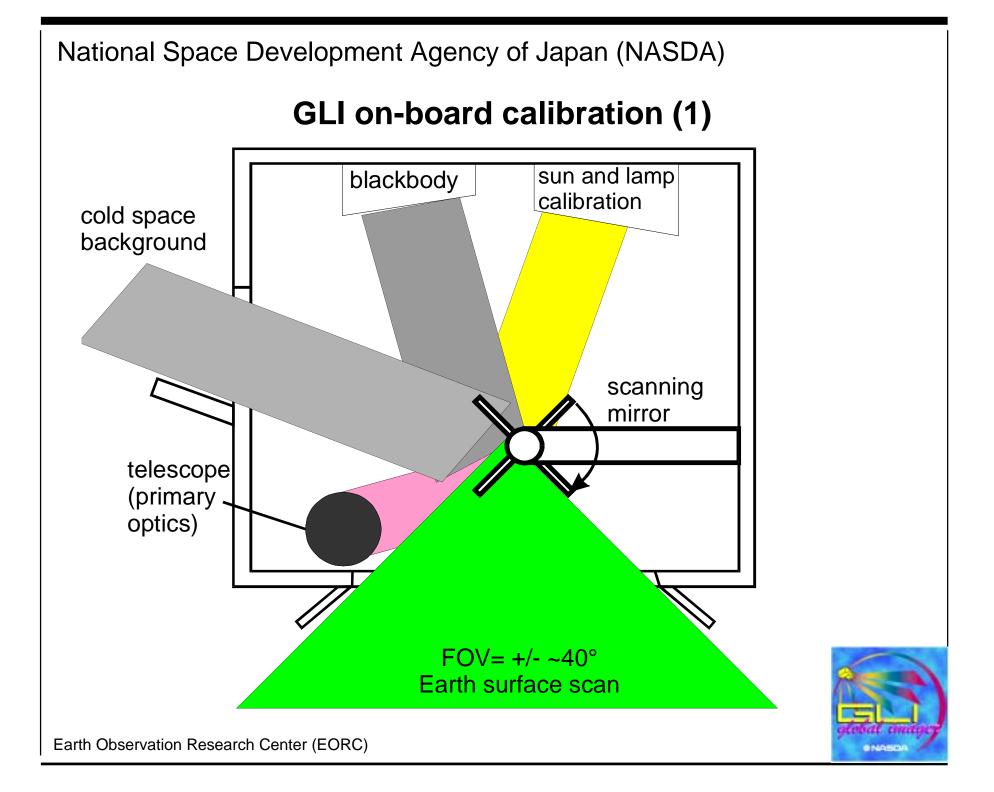
Lessons learnt:

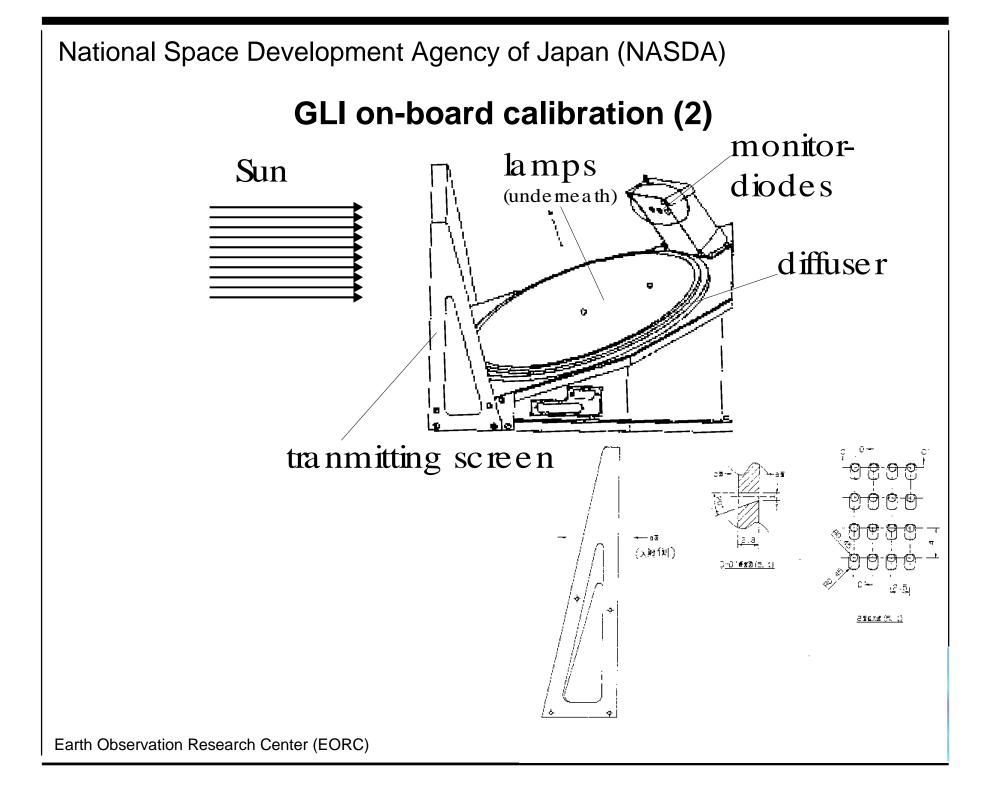
- accurate knowledge of the orbit and illumination geometry is very important for accuracy of the Sun calibration
- more attention to stray light: reduction by a more symmetrical baffle system with respect to the incoming Sun radiation
- different signal levels of Sun calibration (grey diffuser)

Sources: Sümnich & Schwarzer private communication Sümnich & Schwarzer; In-flight calibration of the MOS; Int.J.Remote Sensing (1998)





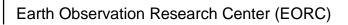




National Space Development Agency of Japan (NASDA)			
Comparison GLI — MOS-IRS			
Sun:	GLI Reduction of solar illumination intensity by a 	 MOS-IRS Direct Sun illumination Shutter protects diffuser 	
Lamps:	 transmitting screen (plate with many holes) Monitor diodes SunCal every orbit LampCal frequently One signal level for each lamp => no linearity measurements Monitor diodes 	 when SunCal is not required LampCal every orbit 2 lamps with different signa levels ensure relative spectral/ radiometric cal. and linearity measurements No thermal channels 	
Thermal:	 Blackbody and cold space 	Excellent results	
Results:	Difficult to forecast		

wheth counts

• NASDA



Conclusion

The high potential of on-board calibration has been shown

- a very high calibration accuracy can be achieved
- in contrast to vicarious calibration, on-board calibration is only technology related

Possible problems for GLI on-board calibration

- possible straylight during Sun calibration
- degradation and/or contamination of the lamps, diffuser and/or transmitting screen

More effort on on-board calibration such as

- straylight measurements / simulation (SunCal system)
- degradation measurements / simulation (diffuser, lamps)



After launch calibration of EO Sensors

Why?

possible sensitivity change of channels during the mission

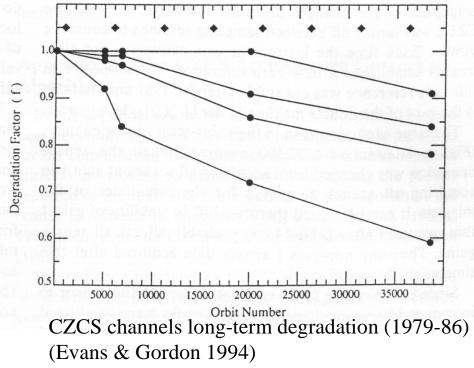
- 5 7 years mission duration
- space environment factors

How accurate?

by **in-flight** calibration: $\epsilon \sim 1\%$

mainly driven by

- design parameters
- technology related



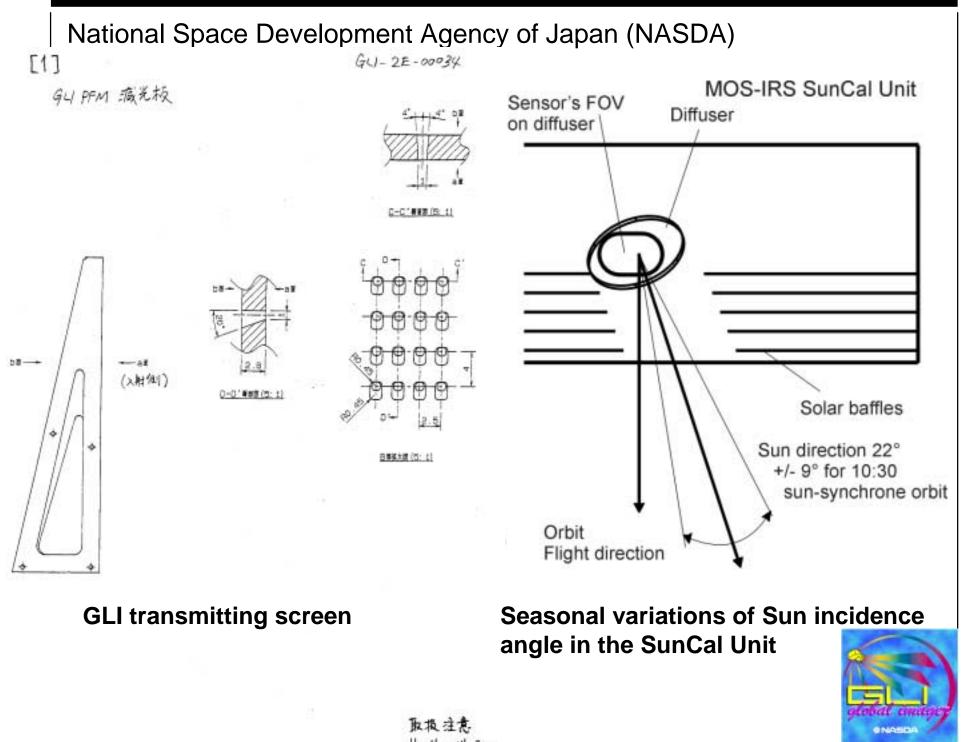
by **vicarious** calibration:

ε **> 3%**

mainly driven by

- dependency on other radiometers
- TOA modelling accuracy





Handle with Care

